

(19)



Europäisch Pat entamt
European Pat ent Offic
Office européen des bre v ts



(11)

EP 0 992 721 A1

(12)

EUROPEAN PATENT APPLICATION

(43) Date of publication:
12.04.2000 Bulletin 2000/15

(51) Int Cl. 7: F16J 15/32

(21) Application number: 99307871.6

(22) Date of filing: 06.10.1999

(84) Designated Contracting States:
AT BE CH CY DE DK ES FI FR GB GR IE IT LI LU
MC NL PT SE
Designated Extension States:
AL LT LV MK RO SI

(30) Priority: 08.10.1998 GB 9821927

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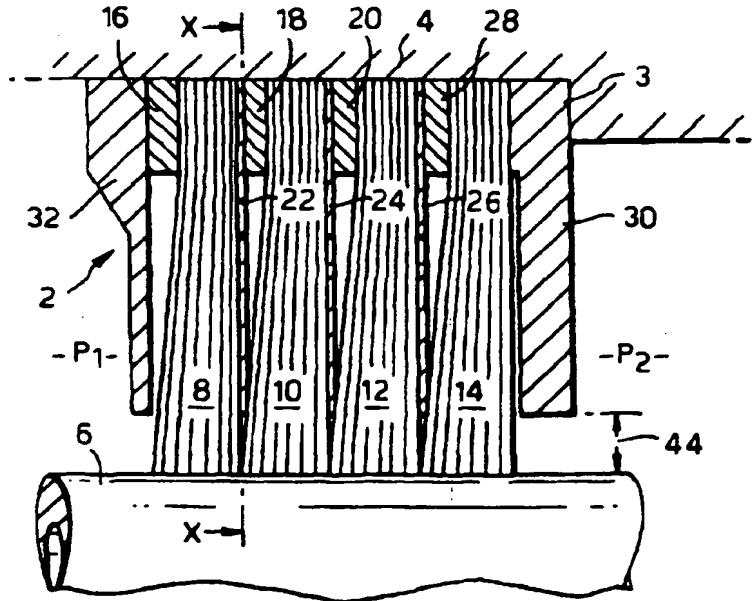
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(54) Improved brush seal

(57) A brush seal (2) designed for use between regions of high pressure differential and in an environment in which there is disturbed flow has a bristle pack subdivided into a plurality of bristle pack regions (8,10,12,14) with a leaky diaphragm (22) between each region. The brush seal (2) may comprise an assembly of individual seals (8-14) with diaphragms (22-26) between them or a single bristle pack which is sub-divided

by at least one internal diaphragm (22). The diaphragms (22-26) are preferably compliant and designed to allow a predetermined leakage flow between bristle regions. The leakage flow has several functions: it causes the pressure differential across the whole seal assembly to be more evenly distributed, and by providing an amount of flow in each pack this acts to calm the bristles thus helping to ensure each bristle region effectively contributes to the seal performance.

Fig.1.



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Description

[0001] The invention concerns the design of an improved brush seal. In particular, the invention relates to a brush seal in which the bristle pack is, effectively, subdivided into upstream and downstream portions by at least one annular diaphragm.

[0002] The basic configuration of a brush seal comprising a simple bristle layer, even with a deep backing plate, does not cope well with high pressure differentials and in regions of disturbed swirling airflow on the high pressure side, for example. A number of factors seem to contribute to poor seal performance in these conditions. On one hand, conditions of high pressure difference causes the bristles to become locked together so that they behave more like a solid mass. When this occurs the bristles are slow to track relative lateral movements of the component against which the seal is produced, for example a rotating shaft, so that once pushed aside by a transient movement the bristles lock-up leaving an increased leakage gap. If the bristles become locked in their undeflected positions initial leakage is low but the tips of the bristles rapidly wear against the co-operating seal surface on the shaft, and the wear rate is increased by shaft excursions, irregularities etc. On the other hand, in disturbed conditions the bristles are caused to flutter with detrimental effect on their co-operative ability to seal effectively because the bristles become permanently splayed destroying their seal properties. Furthermore the deterioration begins with the bristles at the exposed surface and progresses deeper into the layer. Initially the seal appears to perform as expected because the sealing function falls upon bristles further into the layer until eventually all the bristles are corrupted. These two effects may be combined so that while exposed bristles are disturbed and becoming permanently splayed the burden of sealing a high pressure differential falls on bristles towards the back of the layer which are locked together and subject to rapid wear.

[0003] Previous attempts to solve these problems, but without long lasting success, have involved assemblies of multiple brush seals. However, it has been found that under dynamic conditions several stages of adjacent brush seal do not contribute equally and a major proportion of the pressure drop across the whole seal assembly is borne by a single stage. Where turbulent conditions prevail the first, upstream seal has a calming influence on the leakage flow and the second or final stage bears most of the pressure drop. The said one stage gives about equal performance to a single stage seal and overall almost nothing is gained in the long term.

[0004] The present invention seeks to provide a multi-stage brush seal having improved performance by forcing a more even distribution of pressure drop across the individual stages simultaneously creating bristle calming conditions..

[0005] Therefore, according to the present invention,

there is provided a multi-stage brush seal assembly for an annular leakage gap between regions of high pressure differential comprising a plurality of close-coupled bristle pack regions positioned one adjacent another in flow sequence, with partially compliant and leaky diaphragm means located between at least two adjacent bristle pack regions to act as a barrier therebetween and which, in use, provides a controlled low-level flow between the pack regions.

[0006] Each of the close-coupled bristle pack regions may comprise, alternatively, the bristle pack of an individual brush seal or a layer of a single, sub-divided bristle pack wherein a leaky diaphragm is interposed between adjacent bristle pack layers. Furthermore the leaky diaphragm means may be realised in one of several ways. For example, the diaphragm means may comprise an annular diaphragm consisting of a multiplicity of diaphragm segments spaced circumferentially around a common support annulus. The segmented diaphragm and support annulus may be formed integrally from a single sheet of flexible material. Alternatively, the leaky diaphragm may comprise a further thin layer of bristles which may be of different size to the bristles of the pack region, the bristles of the thin layer may be disposed at a different angle or even in a different direction relative to the bristles of the pack regions.

[0007] The invention will now be described in greater detail with particular reference to the embodiments illustrated, by way of example only, in the accompanying drawings, in which:

Figure 1 shows a section through a first embodiment of a brush seal assembly in accordance with the invention comprising a plurality of individual brush seals with a leaky diaphragm between adjacent seals,

Figure 2 shows a similar section through a second embodiment comprising a single bristle pack subdivided by internal leaky diaphragms;

Figure 3 shows in plan view a portion of the leaky diaphragm used in the embodiments of Figure 1 and Figure 2 comprising a segmented plate, and

Figure 4 shows a corresponding view of a further embodiment in which the leaky diaphragm comprises a differentially laid layer of bristles.

[0008] Brush seals are generally useful for effecting a seal between a housing and a shaft mounted for rotation with respect to the housing at potentially very high speeds. A typical application is in gas turbine engines where a brush seal may replace a mechanical seal such as a knife edge or labyrinth seal, but they are also found in other applications. The brush seal is useful to provide a controlled leakage seal, that is a partial barrier, in a gap between regions of different pressures. Excessive

or uncontrolled gas leakage in a gas turbine engine is undesirable for a number of reasons: for example it reduces engine efficiency through thermodynamic cycle losses, and in an internal air cooling system excessive leakage can result in loss of pressure head to drive the cooling flow.

[0009] The design and construction of a typical brush seal, its orientation and manner of fitting are well known in the art and at least a basic knowledge of them will be assumed as background for the purposes this description. The sealing effect is produced by closely packed bristles in a bristle layer co-operating to provide a diffuse leakage path of very low overall leakage rate. Providing the integrity of the bristle layer is maintained its sealing function remains at a high level then brush seal leakage rates are appreciably lower compared to labyrinth seals. This offers significant gains in engine performance due to the reduction of losses in the internal airflow system. The typical seal design has a layer of densely packed fine wire bristles sandwiched between a pair of backing plates. The downstream backing plate extends almost the full radial depth of the bristles and its clearance over the shaft defines the actual leakage gap. Its clearance is large enough to accommodate lateral shaft movements, differential thermal growth, shaft orbiting etc. An upstream plate is not always utilised but may be useful in some instances for calming bristle flutter and blow down.

[0010] In practice difficulties arise in maintaining the integrity of a seal under all conditions. Aerodynamic conditions in the immediate vicinity of a seal, in particular high pressure differentials and turbulent flow regimes, have been found to detrimentally affect seal performance. The effect on single layer seal stages is especially marked. Multiple seal cascades often fail to deliver an expected improvement and this has been found to be the result of several factors. On one hand high pressure differentials can cause the bristles to lock together tending to cut-off the inter-bristle diffuse flow so that the bristle pack effectively becomes a solid mass. The inherent resilience of the individual bristles is thereby lost so that during excursions the shaft impacts the bristles which suffer accelerated wear as a result. Also they may be deflected sideways and be unable to return to their sealing position. The bristles may be blown-down into contact with the shaft leading to premature wear. On the other hand turbulent flow in front of the seal bristle layer can also destroy the effectiveness of the seal by causing the bristles to flutter so that they cease to co-operate in providing the diffuse flow path. Both conditions have been observed to occur simultaneously in one seal with bristles at the surface of the pack in a state of flutter while deeper bristles are locked together. Our earlier patent GB 2,281,108B addressed this particular problem by means of a porous cover or shield over the front of the bristle pack to restore benign conditions at the pack face returning a diffuse flow through the bristles and calming their behaviour.

[0011] Multiple seal assemblies are seen to suffer the same problems with the additional difficulty that the total pressure drop is not equally shared by all seal stages. Test rig measurements indicate that the most downstream seal stage carries a significantly greater proportion of any given pressure drop and tends to perform in the manner of a single seal stage and is susceptible to the drawbacks discussed above. The present invention attempts to solve these problems in a multiple seal stage

5 arrangement by distributing a pressure differential more evenly between the several stages while ensuring the bristles of each stage remain effective.

[0012] Referring now to Figure 1, there is shown in accordance with the present invention a multi-stage 10 brush seal assembly at 2, located against an annular shoulder 3 in a housing 4 to control flow through an annular leakage gap 5 between the housing and a concentric, relatively rotatable rotor 6. A region of relatively high pressure is represented at P₁ to the left and upstream 15 side of the illustration and a region of substantially lower pressure is represented at P₂ on the right and downstream side of the illustration. Therefore leakage flow from P₁ to P₂ through the seal 2 and the gap 5 is from left to right in the drawing in Figure 1. The same convention is also used in Figure 2.

[0013] The particular brush seal assembly 2 illustrated in Figure 1 comprises four stages of close-coupled bristle pack regions, indicated at 8, 10, 12, 14 from the left or upstream side. In this embodiment the bristle pack 20 regions are formed as four separate layers each consisting of an individual brush seal, although they may be assembled and packaged as a single unit. The construction of each of the bristle pack regions can be compared to a conventional single stage brush seal, insofar as each has an upstream ring and a downstream backing ring with a bristle pack layer sandwiched therebetween. They also follow normal practice in construction in that the bristle pack and the two rings are welded together. Thus, in Figure 1 the seal assembly includes from the 25 left upstream rings 16, 18, 20 and 28, and downstream backing plates 22, 24, 26, 30. An upstream protective cover or shield, of known configuration, is also disposed abutting the upstream side of ring 16 adjacent the upstream bristle pack 8. The right most downstream backing ring 30 is also of known configuration. Conventionally it is solid and rigid and functions as an air dam in the leakage path.

[0014] In accordance with the present invention the intermediate backing plates 22, 24, 26 interposed between the adjacent upstream and downstream faces of the bristle pack regions 8, 10, 12, 14 comprise partially compliant and leaking diaphragm means, in direct contrast to a conventional backing plate, such as at 30.

[0015] The number of bristle pack regions or layers mentioned, i.e. four, is by way of example only and is not intended to be recommended as an optimum number nor as possessing any specially desirable or preferred characteristic.

[0016] The second embodiment illustrated in Figure 2 is essentially equivalent to the first embodiment but, whereas the first embodiment comprises an assembly of several individual brush seals, in the second embodiment the several seal stages are formed integrally as a single bristle pack with internal sub-division. With reference to the drawings in order to emphasise correspondence between the constituent parts of the two embodiments like parts are given like reference numerals in Figures 1 and 2. Thus, the four annular bristle packs 8,10,12,14 are located one up against another with only the thin backing plates or diaphragms 22,24,26 between them. This composite assembly is sandwiched between rings 34,36 on its upstream and downstream sides respectively. Although these rings 34,36 are shown in the drawing as simple rings of limited radial depth they may be replaced respectively by the front shield 32 and the backing plate 30 of substantially greater depth, as employed in the first embodiment illustrated in Figure 1.

[0017] As will now be described the thin backing plates or diaphragms 22,24,26 lying intermediate the bristle packs are formed to be intentionally leaky. That is, the diaphragms do not simply act as air dams but are also pierced at mid-height to provide pressure equalisation and a controlled leakage path. A detail view on the section X-X of Figure 1 is shown in Figure 3, this view is common to all the thin backing plates or diaphragms 22,24,26 intermediate the bristle layers in both embodiments. The diaphragm 22, in common with the others, is annular and made of a relatively thin sheet, the material of these sheets may be a metal or metal alloy, such as stainless steel or nickel alloy to withstand the high ambient temperatures encountered. The diameter of the outer annular circumference 40 is equal to the outer circumference of the seal and extends radially inwards for almost the full depth of the bristle layers to an inner annular circumference 42. The dimension of the seal gap 44 between the inner circumference 42 of the annular diaphragm 22 and an outer cylindrical surface 46 of the rotor 6 is chosen according to the same criteria as the corresponding gap between a deep backing plate 30 or a deep front shield 32 and the rotor.

[0018] The design of the diaphragm 22 is intended to serve two functions which to some extent at least are inter-related. By sub-dividing the overall bristle pack it is intended to alleviate the drawbacks or deficiencies in sealing performance previously noted by introducing measures to distribute the overall pressure drop more evenly between the sub-layers and also to introduce a preferential flow in the bristle pack which will tend to calm disturbed bristles.

[0019] A certain amount of compliance in the diaphragms is provided by partial segmentation. The inner circumferential margin of the diaphragm 22, that is an annular region adjacent the inner circumference 42 is divided into a multiplicity of circumferentially extending segments to which the reference 48 is allotted and several of which are visible in the drawing. These segments

are formed by a multiplicity of radially extending slots 50, preferably of a "keyhole" shape which extend in radially outward directions from, and are spaced apart around the inner circumference 42. The slots 50 extend

5 radially outwards for a substantial proportion of the radial depth of the bristle layer so that each segment is independently supported by an outer annular region 52 of the diaphragm. Each of the slots 50 is terminated by a circular hole 54. The primary function of these holes 10 is to provide a controlled leakage path through the diaphragm into a region of the adjacent bristle pack close to the proximal end of the bristles, but they also function as stress relief features.

[0020] Leakage flow through the holes 54 will tend to 15 flow generally in a radial direction along the length of the bristles, towards their free ends, and as has been observed in the prior art such flows have been found to have a calming influence on disturbed bristles. The diameter of the holes 54 is selected to provide a desired 20 leakage rate between bristle layers. The stress relief secondary function operates to prevent crack propagation from the ends of the slots 50 into the remainder of the diaphragm 22.

[0021] Also the slightly reduced distance between 25 neighbouring holes 54 tends to encourage a preferential bending zone for each segment. Individual segments are therefore relatively movable, to a limited extent, enabling the diaphragms to be partially compliant and to allow some axial bending of the bristles while tending to 30 damp the movement.

[0022] In the drawing of Figure 3 the bristles in the 35 layer behind the diaphragm 22 are visible in the gap 44 and through the slots 50 and through the leakage holes 52. As indicated in Figures 1 and 2 these bristles belong to the bristle pack region 10. As is conventional in the art the bristles are disposed at an angle to the rotor surface 46 usually about 45°. The presence of a multiplicity of radially extending slots in the diaphragm together with the multiplicity of holes determines the rate at which the 40 diaphragm will leak airflow into the downstream bristle layer over a range of depths rather than simply acting as a weir or dam, as would a solid diaphragm. Thus a diffuse air flow is promoted through the layers of the bristle pack in a direction having a radial component. The 45 effect of this flow is to tend to calm disturbed bristles in turbulent conditions and to tend to release bristles which are otherwise locked together. Furthermore, the individual segments of the diaphragm possess a degree of independent compliance, since the region of reduced segment width between neighbouring holes tends to create 50 a preferred region of bending for individual segments. As the diaphragms are disposed in planes perpendicular to the common axis of the brush seal and the shaft, bending takes place in an axial direction so that this 55 movement is available to relieve the pressure induced axial forces acting on the bristles which tend to cause them to lock together. Thus, the compliance of the diaphragm segments in conjunction with the induced radial

leakage flow act cooperatively to reduce bristle lock up thereby releasing intermediate portions of the bristle pack to track excursions of the rotor. As a result the sealing properties of the pressure gradient through the seal length is more evenly distributed and overall seal performance is improved.

[0023] Referring now to Figure 4, there is shown an alternative form of realisation of partially compliant and leaky diaphragm means. For ease of comparison with the embodiment illustrated in Figure 3 like parts have been given like references. Thus, there is shown housing 4, shaft 6 and bristle pack region 10. The diaphragm means generally indicated at 22 comprises a further annular layer of bristle 56. Preferably the bristles 56 are of a different size, thicker, than the bristles of the adjacent bristle pack regions 8,10; and are disposed at a different lay angle. It has been found advantageous to arrange the bristles 56 in a radial direction with their radially inner tips (in the illustrated example) spaced a short distance from the surface of the rotor, as in the instance of the inner circumference of the diaphragm 52 of Figure 3.

[0024] In one alternative configurations the bristles 56 may be simply laid at a different angle, but sloped in the same direction as the bristles 10. In another configuration the bristles 56 are sloped in the opposite direction to the bristles 10 in the adjacent pack regions. In all cases the bristles 56 lay across the bristles of the adjacent regions and are thus able to support the bristles and resist axial movement. However, like all bristle regions a certain level of diffuse flow passes through the bristle pack, thus passing from the downstream face of the upstream bristle pack, through the diaphragm layer, and into the upstream face of the downstream bristle pack. The resulting effect on bristle behaviour is therefore substantially the same as that achieved with the diaphragm means 52 of the Figure 3 embodiment.

Claims

1. A multi-stage brush seal assembly for an annular leakage gap between regions of high pressure differential comprises a plurality of bristle pack regions 8,10,12,14 positioned one adjacent another in flow sequence, with partially compliant and leaky diaphragm means 22,24,26 located between at least two adjacent bristle pack regions which, in use, provides a controlled low-level flow between the pack regions, characterised in that the plurality of bristle pack regions 8,10,12,14 are close-coupled one adjacent another and the intervening leaky diaphragm 22,24,26 means is at least partially compliant.
2. A multi-stage brush seal assembly as claimed in claim 1 further characterised in that the close-coupled bristle pack regions each comprises an individual brush seal (Figure 1) located so as the upstream face of a downstream seal lies adjacent the down-
5. stream face of an upstream seal and leaky diaphragm means is interposed the between confronting upstream and downstream faces of adjacent brush seals.
3. A multi-stage brush seal assembly as claimed in claim 2 wherein an upstream face shield 32 is located adjacent the upstream face of the upstream bristle pack 8 and a downstream backing plate 30 is located adjacent the downstream face of the downstream bristle pack 14 further characterised in that the face of the intermediate bristle packs 10,12 have neither backing plate nor face plate.
10. 4. A multi-stage brush seal assembly as claimed in claim 1 further characterised in that the plurality of close-coupled bristle pack regions 8,10,12,14 comprises a single bristle pack (Figure 2) internally subdivided by at least one partially compliant and leaky diaphragm means.
15. 5. A multi-stage brush seal as claimed in any preceding claim further characterised in that the or each leaky diaphragm means 22,24,26 (Figure 3) comprises an annulus 52 of relatively stiff material compared to the stiffness of the pack bristles 10.
20. 6. A multi-stage brush seal assembly as claimed in claim 5 further characterised in that the or each leaky diaphragm means 22 comprises an annular plate 52 of stiff material formed with a multiplicity of slots 50 extending from one annulus circumference in a generally radial direction, said slots being spaced apart around the circumference whereby to form a multiplicity of independently movable segments 48.
25. 7. A multi-stage brush seal as claimed in claim 6 wherein the slots 50 extend for a substantial part of a radial depth of the diaphragm 52.
30. 8. A multi-stage brush seal as claimed in either claim 6 or claim 7 wherein the slots 50 are of narrow width sufficient only to permit relative movement of adjacent segments 48.
35. 9. A multi-stage brush seal as claimed in any one of claims 4 to 6 wherein each slot 50 is terminated by a small hole 54 of diameter slightly greater than the width of a slot 50, and the diameter of the hole 54 is selected according to a required leakage flow.
40. 10. A multi-stage brush seal assembly as claimed in claim 5 further characterised in that the or each leaky diaphragm means 22 (Figure 4) comprises a further annular layer 56 of bristles.
45. 11. A multi-stage brush seal assembly as claimed in

claim 10 wherein the further layer of bristles 56 comprises wires of a different size compared to the wires of an adjacent bristle pack region 10.

12. A multi-stage brush seal assembly as claimed in claim 10 or claim 11 wherein the bristles of the further layer 56 are disposed at a different lay angle compared to the lay angle of the bristles of an adjacent bristle pack region 10. 5

13. A multi-stage brush seal assembly as claimed in any of claims 10 to 12 wherein the bristles of the further layer 56 are disposed in a different direction compared to the bristles of an adjacent bristle pack region 10. 10 15

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Fig.1.

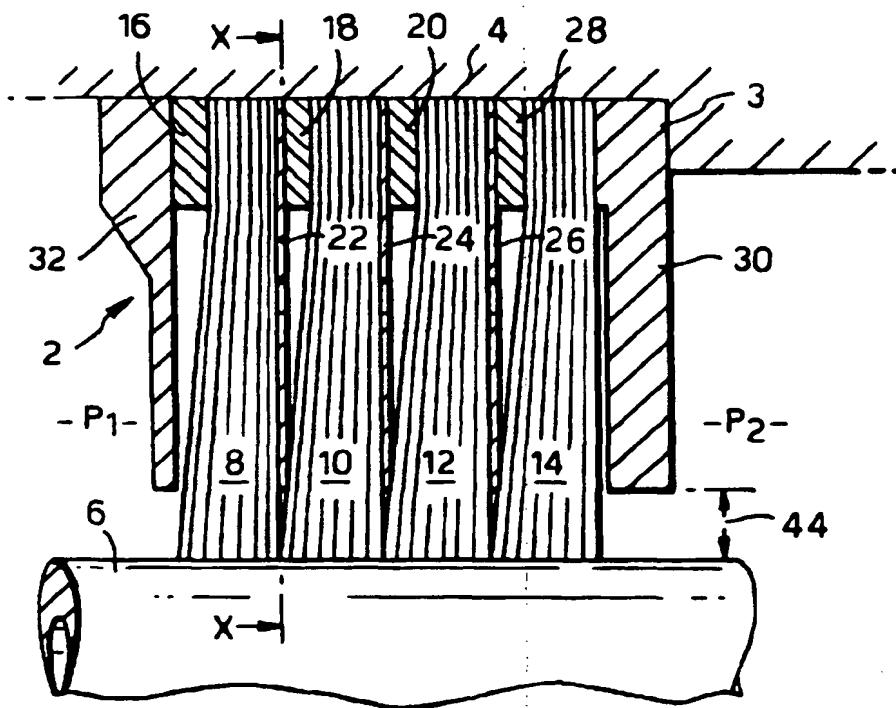


Fig.2.

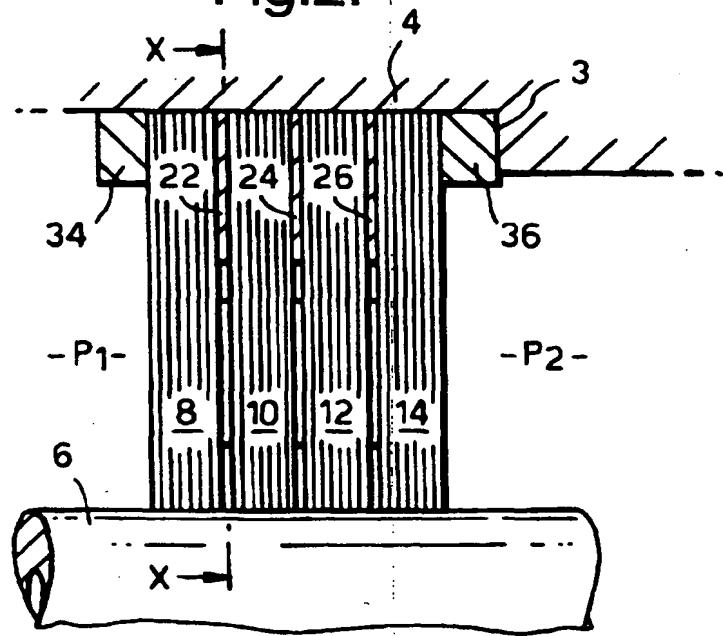


Fig.3.

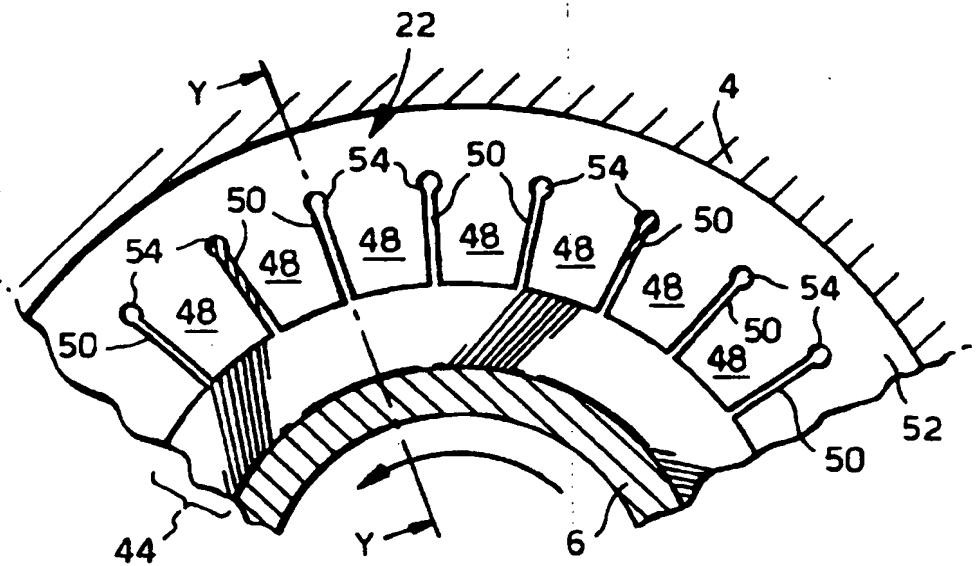
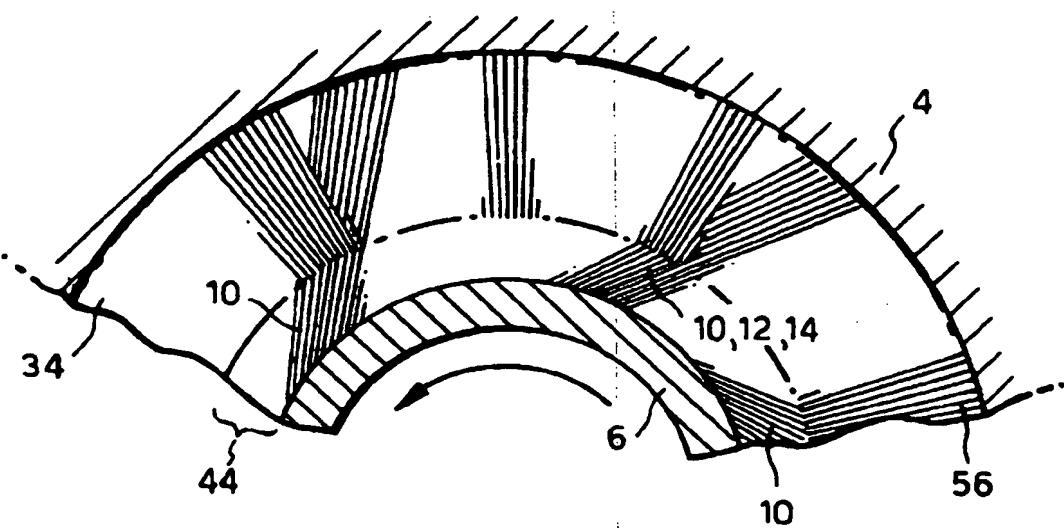


Fig.4.





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EUROPEAN SEARCH REPORT

Application Number

EP 99 30 7871

DOCUMENTS CONSIDERED TO BE RELEVANT			CLASSIFICATION OF THE APPLICATION (Int.Cl.7)						
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim							
Y	US 5 106 104 A (ATKINSON EDWARD ET AL) 21 April 1992 (1992-04-21) * the whole document *	1	F16J15/32						
A	---	2							
Y	US 5 568 931 A (TSENG WU-YANG ET AL) 29 October 1996 (1996-10-29) * the whole document *	1							
A	---	5-9							
A	WO 92 14951 A (CROSS MFG CO) 3 September 1992 (1992-09-03) * page 7, line 19 - page 9, line 35; figures 3-5 *	3,10-13							
A	US 5 474 306 A (BAGEPALLI BHARAT S ET AL) 12 December 1995 (1995-12-12) * column 3, line 14 - line 55; figures 3,4 *	4							
A	WO 92 05378 A (CROSS MFG CO) 2 April 1992 (1992-04-02) * page 1; figures 1,6 *	1,13	TECHNICAL FIELDS SEARCHED (Int.Cl.7)						
A	WO 97 14897 A (SEALOL) 24 April 1997 (1997-04-24) * page 1; figure 1 *	1	F16J F01D						
<p>The present search report has been drawn up for all claims</p> <table border="1" style="width: 100%; border-collapse: collapse;"> <tr> <td style="width: 33%;">Place of search</td> <td style="width: 33%;">Date of completion of the search</td> <td style="width: 34%;">Examiner</td> </tr> <tr> <td>BERLIN</td> <td>20 January 2000</td> <td>Hoffmann, M</td> </tr> </table>				Place of search	Date of completion of the search	Examiner	BERLIN	20 January 2000	Hoffmann, M
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BERLIN	20 January 2000	Hoffmann, M							
CATEGORY OF CITED DOCUMENTS		T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document							
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ON EUROPEAN PATENT APPLICATION NO.**

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This annex lists the patent family members relating to the patent documents cited in the above-mentioned European search report. The members are as contained in the European Patent Office EDP file on The European Patent Office is in no way liable for these particulars which are merely given for the purpose of information.

20-01-2000

Patent document cited in search report		Publication date	Patent family member(s)			Publication date
US 5106104	A	21-04-1992	NONE			
US 5568931	A	29-10-1996	NONE			
WO 9214951	A	03-09-1992	US	5480165 A		02-01-1996
US 5474306	A	12-12-1995	US	5749584 A		12-05-1998
WO 9205378	A	02-04-1992	DE	69121244 D		12-09-1996
			DE	69121244 T		20-03-1997
			EP	0549652 A		07-07-1993
			JP	6501298 T		10-02-1994
			US	5474305 A		12-12-1995
WO 9714897	A	24-04-1997	EP	0850373 A		01-07-1998
			JP	11513781 T		24-11-1999